# Calculation Procedures for Shear Stress vs Shear Strain

NOTATIONS (Unit system : N / mm<sup>2</sup> = MPa = 0.0981 kgf / cm<sup>2</sup>, concerning only for calculating  $f_{cr}$  from  $f_c$ )

	t	: panel thickness	カウントライン
	$ au_{xy}$	: shear stress	
	$f_y$	: y-direction stress (=0)	XXX
	$f_x$	: x-direction stress (=0)	$\mathbf{K}\mathbf{X}\mathbf{X}$
	$\gamma_{xy}$	: shear strain	
	S <sub>m</sub>	: average crack interval (45 degree direction)	
		$=(150+250+150+250)\sqrt{2}$	
		/ number of cracks on count lines	
	$\mathcal{E}_1$	: tensile principal strain	
	$\mathcal{E}_2$	: compressive principal strain	le Le
	$\mathcal{E}_{x}$	: x-direction strain	
	$\boldsymbol{\mathcal{E}}_{y}$	: y-direction strain	
	$\theta$	: angle between x-direction and compressive	A I fri
<i-th reinforcemen<="" td=""><td><math>i \to (i-1)</math> to</td><td>principal strain (clockwise as positive) n <math>n = &lt; 10</math> is available)</td><td>ri-</td></i-th>	$i \to (i-1)$ to	principal strain (clockwise as positive) n $n = < 10$ is available)	ri-
	$E_{ii}$	: elastic modulus	xy fc1
	$f_{rui}$	: tensile strength or yield strength	$\leftarrow$
	$f_{ri}$	: average stress	,
	${\cal E}_{ri}$	: average strain	
	$A_{ri}$	: sectional area	
	S <sub>i</sub>	: interval (pitch)	
	$ ho_i$	: sectional area ratio = $\frac{A_{ri}}{s_i \cdot t}$	
	$ heta_{\scriptscriptstyle ri}$	: angle between x-direction and i-th reinforcement	
		(clockwise as positive $0^{\circ} \le \theta_{ri} < 180^{\circ}$ )	
	$ heta_i$	: angle between tensile principal strain and i-th reinforce	ement
		(clockwise as positive = $(\theta + 90^\circ) - \theta_{ri}$ )	
	$ ho_{_{ix}}$	: effective ratio for x-direction = $\rho_i  \cos \theta_{i} $	
	$ ho_{_{iy}}$	: effective ratio for y-direction = $\rho_i \sin \theta_{ri}$	
	$P_{rcri}$	: tensile force at crack	
	$P_{rmi}$	: average force of reinforcement	
	$f_{rcri}$	: tensile stress at crack	
<concrete></concrete>			
	${f}_{c}$	: compressive strength (negative value)	
	$\mathcal{E}_c$	: strain at compressive strength (negative value)	
	$f_{cr}$	: crack strength	

 $\mathcal{E}_{cr}$  : strain at crack strength

$E_{c}$	: elastic modulus (calculating as $2 \cdot f_c / \varepsilon_c$ )
$f_{c1}$	: average tensile stress
$f_{c2}$	: average compressive stress
$f_{c2\max}$	: compressive strength of cracked concrete (negative value)
$n_i$	: elastic modulus ratio of reinforcement to concrete $= E_{ii}$ /

<others>

: yield bond stress  $\tau_{bv}$ 

 $s_{by}$ : slippage at yield bond stress

### Step 1

Give  $\varepsilon_1$  for calculating this procedure.

# Step 2 Give $\theta$ in arbitrary.

Step 3 Give  $\Sigma f_{riv}$  in arbitrary.

#### Step 4

Calculate  $f_{c1}$  (Modified Compression-Field Theory by Collins et al.).

$$\begin{cases} f_{c1} = \frac{2 \cdot f_c \cdot \varepsilon_1}{\varepsilon_c} \quad (\varepsilon_1 \le \varepsilon_{cr}) \\ f_{c1} = \frac{f_{cr}}{1 + \sqrt{200 \cdot \varepsilon_1}} \quad (\varepsilon_1 \ge \varepsilon_{cr}) \\ f_{cr} = 0.33\sqrt{-f_c} \quad \varepsilon_{cr} = \frac{f_{cr} \cdot \varepsilon_c}{2 \cdot f_c} \end{cases}$$

Step 5

Calculate  $au_{xy}$ .

$$\tau_{xy} = \frac{f_{c1} + \Sigma \rho_{iy} \cdot \Sigma f_{riy}}{\tan \theta}$$

Step 6

Calculate  $f_{c2}$ .

$$f_{c2} = f_{c1} - \tau_{xy} (\tan \theta + 1 / \tan \theta)$$

Step 7

Calculate  $f_{c2 \text{ max}}$  from  $\varepsilon_1$  (Kanakubo et al., 2000).

$$\frac{f_{c2\max}}{f_c} = -\frac{1}{0.95 \cdot \varepsilon_1 / \varepsilon_c} \le 1.0$$

Step 8

Judge for concrete failure.

If  $f_{c2} > f_{c2\max}$ , finish the procedure.

## Step 9

Calculate  $\varepsilon_2$  (Modified Compression-Field Theory by Collins et al.).



Mohr's stress circle





 $/E_c$ 

$$\varepsilon_{2} = \varepsilon_{c} \cdot \left(1 - \sqrt{1 - \frac{f_{c2}}{f_{c2 \max}}}\right) \qquad \qquad \because \frac{f_{c2}}{f_{c2 \max}} = 2\left(\frac{\varepsilon_{2}}{\varepsilon_{c}}\right) - \left(\frac{\varepsilon_{2}}{\varepsilon_{c}}\right)^{2}$$

Step 10 Calculate  $\varepsilon_y$ .

$$\varepsilon_{y} = \frac{\varepsilon_{1} + \varepsilon_{2} \cdot \tan^{2} \theta}{1 + \tan^{2} \theta}$$

From Step 11 to 14, for each reinforcement (i=1 to n) Step 11 Calculate  $\theta_i$ .

$$\theta_i = (\theta + 90^\circ) - \theta_{ri}$$

Step 12

Calculate  $\mathcal{E}_{ri}$ 

$$\varepsilon_{ri} = \frac{\varepsilon_1 + \varepsilon_2}{2} + \frac{\varepsilon_1 - \varepsilon_2}{2} \cdot \cos 2\theta_i$$

Step 13

Calculate  $f_{ri}$ .

$$f_{ri} = E_{ri} \cdot \varepsilon_{ri}$$
 If  $f_{ri} > f_{rui}$  in case of yield,  $f_{ri} = f_{rui}$ 

Step 14

Calculate 
$$f_{riy}$$
 and  $f_{rix}$ .  

$$\begin{cases}
f_{riy} = f_{ri} \cdot \sin \theta_{ri} \\
f_{rix} = f_{ri} \cdot |\cos \theta_{ri}|
\end{cases}$$

Step 15 Calculate  $\Sigma f_{riy}$  and  $\Sigma f_{rix}$ .

Step 16

Compare  $\Sigma f_{riy}$  between Step 3 and Step 15. If these are not equal, give another  $\Sigma f_{riy}$  and return Step3.

Step 17 Calculate  $\mathcal{E}_x$ .

$$\varepsilon_x = \varepsilon_1 + \varepsilon_2 - \varepsilon_y$$

Step 18 Calculate  $f_x$ .

$$f_{cx} = f_{c1} - \frac{\tau_{xy}}{\tan \theta}$$
$$f_{x} = f_{cx} + \Sigma \rho_{ix} \cdot \Sigma f_{rix}$$

Step 19

Check  $f_x = 0$ . If  $f_x \neq 0$ , then give another  $\theta$  and return Step 2.







Mohr's strain circle

From Step 20 to 24, for each reinforcement (i=1 to n) Step 20 Calculate  $s_{mi}$ .

$$s_{mi} = s_m \frac{\sin\left(135^\circ - \theta\right)}{\left|\cos\theta_i\right|}$$

However, in case of  $\theta_i = 90^\circ$ , no calculating needs.

Step 21 Calculate  $P_{rcri}$ .

$$P_{rcri} = \frac{(1+n_i\rho_i) \cdot P_{rmi} \cdot s_{mi}}{n_i\rho_i \cdot s_{mi} + \frac{2}{k \cdot \tanh(k \cdot s_{mi}/2)}}$$
$$P_{rmi} = f_{ri} \cdot A_{ri} \qquad k = \sqrt{\frac{(1+n_i\rho_i) \cdot \tau_{by}}{E_{ri} \cdot A_{ri} \cdot s_{by}}}$$



Step 22

Calculate  $\tau_{cri}$ .

$$\tau_{cri} = \frac{k \cdot P_{rcri} \cdot \sinh(k \cdot s_{mi} / 2)}{(1 + n_i \rho_i) \cosh(k \cdot s_{mi} / 2)}$$

Step 23

If  $\tau_{cri} \leq \tau_y$ , go to the next step. If  $\tau_{cri} > \tau_y$ , calculate  $P_{rcri}$ .

$$P_{rcri} = \frac{P_{rmi} \cdot s_{mi} + \tau_{y} \cdot {l_{yi}}^{2} + 2E_{ri} \cdot A_{ri} \cdot s_{y} / (1 + n_{i}\rho_{i})}{2l_{yi} + \frac{s_{mi} - 2l_{yi}}{1 + 1/n_{i}\rho_{i}}}$$
$$\frac{P_{rcri} - (1 + n_{i}\rho_{i})\tau_{y} \cdot l_{yi}}{E_{ri} \cdot A_{ri}} + \frac{k \cdot s_{y}}{\tanh(k \cdot s_{mi} / 2 - k \cdot l_{yi})} = 0$$

Step 24

Calculate  $f_{rcri}$ , judge the rupture of reinforcement.

$$f_{rcri} = \frac{P_{rcri}}{A_{ri}}$$
 If  $f_{rcri} > f_{rui}$  in case of rupture, finish the procedure.

Step 25 Calculate  $\gamma_{xy}$ .

$$\gamma_{xy} = \frac{2\left(\varepsilon_x - \varepsilon_2\right)}{\tan\theta}$$

Finishing for one procedure. For next  $\varepsilon_1$ , go to Step 1.